

Utilizing Nano Satellites for Water Monitoring for Nile River

November 23rd, 2013

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Background: Flood and Drought



Background: Water Pollution



Mission Statement

- We aim to establish network for water resource monitoring for Nile river with Hodoyoshi "Store and Forward" (S&F) communication function and an affordable ground sensor (Target total system cost: <\$600).
- Water resource management for Nile river basin will be conducted first.
- After the completion of Nile river project,
 - Water resource monitoring system will be deployed world wide.
 - further development to improve S&F communication network with cubesat-size satellite constellation and to increase versatility of ground sensor will be executed.
 - This project will be an international co-operation project.

Remapping Nile River



Mission Overview (1)

- Water level
- Acidity (pH)
- Clarity (turbidity)
- Oxygen(DO)
- Temperature

Mission Overview (2)

- Sensor selection and the number of sensor will be determined based on:
 - -Sensor availability
 - -Costs
 - -Locations of urgent need
 - -Suggestion of MWRI/NWRC/NRI.

Mission Example: Water Level Monitoring

Observation Range	5
Resolution	0.01
Unit	Meter
Required data bits	9
Observation Frequency per day	24
# of sensors for Nile river coverage	TBD
Sensor availability	Under development
Target sensor cost	<\$100
Note	Observe water level change in short term and long term.

Success Criteria

- Minimum success
 - At least <u>one</u> sensor data from <u>one</u> location is successfully received by a Hodoyoshi satellite and forwarded to a ground station for 6 months.
- Full success
 - <u>All</u> sensor data from <u>one</u> location are successfully received by a Hodoyoshi satellite and forwarded to a ground station for 2 years.
- Extra success
 - <u>All</u> sensor data from <u>multiple</u> locations are successfully received by a Hodoyoshi satellite and forwarded to a ground station for 2 years.

Concept of Global Network for On-Ground Sensors with Nano/Micro Satellites (Application: Water Level Monitoring)

S&F satellite constellation



Space Segment: HODOYOSHI-3 & 4

Hodoyoshi-3



Hodoyoshi-4



	Hodoyoshi-3	Hodoyoshi-4				
Size	0.5 × 0.5 × H0.65m	0.5×0.6×H0.7m				
Weight	60kg	66kg				
Orbit	SSO. 600km, LTAN	10am~11am				
ACS	Earth pointing, 3 ax	is stabilization				
Power	Power generation:	max 100W				
	Power consumptior	n: average 50 W				
	Bus voltage:	28V, 5V				
	Battery:	5.8AH Li-Ion				
Commu-	H/K and Command:	S-band				
nication	uplink:4 kbps, downlink:4/32/64 kbps					
	Mission data downlink: X-band 10Mbps					
	(100Mbps to be tested on Hodoyoshi-4)					
Orbit	H_2O_2 propulsion	lon-thruster				
control		(Isp: 1100s)				
Missions	Mid-resolution	High-resolution optical				
	optical camera	camera				
	GSD:40m & 200m	GSD:5m				
	Store & Forward					
	Hosted payloads (10cm cube x 2)					
	Hetero-constellation experiment					

"Store and Forward" Receiver

Function and Spec				
UHF frequency	400 MHz			
High speed A/D conversion	- AN			
Sampling frequency	10 kHz or 40 kHz			
Sampling time	1 sec or 10 sec			
Modulation (Data transmission)	BPSK			
Data storage capacity	Up to 16 Gbits (nonvolatile memories)			
Digital data transfer speed	Up to 10 Mbps (Target)			
Power supply	Unregulated power bus betweer +16 V and +36 V	า		
Power consumption	Up to 5 W (Target)			
Size	150 mm x 150 mm x 35 mm (excluding fitting mount)			
Development status	FM integration and testing			



Characteristics:
No on-board
demodulation
High-speed
A/D conversion
of received signals.

Water Resource Monitoring Sensor (1)

Original Concept (Water level monitoring)



Water Level Monitoring Sensor (2)

Current Design: Abiki-kun R



Sensor: URM37 V3.2 Ultrasonic Sensor Voltage: 5 V Current: less than 20 mA Observation range: 4 cm – 5 m Interface: TTL or RS232





Water Level Monitoring Sensor (3)

Current Design: Abiki-kun R



Source: Water level monitoring by an ultrasonic distance measuring sensor (Analysis of distance measuring unit for Abiki-kun R), Nagasaki nishi high school, Earth science club

Water Level Monitoring Sensor (3)

Three different configuration were tested.

Configuration	With a 2m vinyl chloride tube	With a float	Measurement method	System feasibility	Note
Original	Yes	Yes	Camera	No	Too much resources required.
Abiki-kun R # 1	Yes	Yes	Ultrasound	Yes	Measurement error is relatively small
Abiki-kun R # 2	Yes	No	Ultrasound	Yes	Measurement stability might be degradated by waves.
Abiki-kun R # 3	No	No	Ultrasound	Yes	Measurement stability might be degradated by waves.

Source: Water level monitoring by an ultrasonic distance measuring sensor (Analysis of distance measuring unit for Abiki-kun R), Nagasaki nishi high school, Earth science club

Store and Forward Transmitter

	Spec
Frequency	400 MHz
Modulation	ASK+BPSK
Bandwidth	Less than 30 kHz
Speed	300 bps
Transmission power	1 W nominal Low power mode (1 μW, 10 mW , 100 mW)
Power consumption	During data transmission: 5 W Stand-by mode: 50 mW Sleep mode: 1 mW
Size	150 mm × 80 mm × 30 mm
Weight	Less than 200g

Development status:

- A prototype transmitter is being manufactured.
- Field testing will be performed by the end of this year.

Store and Forward Transmitter Specifications

Data Transmission (1)

- Data Transmission mode: 1 sec/10 sec
- Data Transmission speed: 300 bps
- Signal recognition and info. header : 0.1 sec
- Transmittable data size per one data tranmission attempt
 - -1 sec mode: 270 bits (0.9 second for data)
 - -10 sec mode: 2970 bits (9.9 seconds for data)

Data Transmission (2)

- Estimation of Hodoyoshi satellite AOS/LOS, and timing of data transmission
 - A sensor keeps orbital elements of Hodoyoshi satellite and estimate AOS/LOS time.
 - -Orbital elements become inaccurate over time
 - -Multiple data transmission attempt between AOS and LOS. (2-4 times, once every minute)

Data Transmission (3)

 Observation data will be sent <u>twice</u> to prevent data transmission failure.



Data Transmission (4)

- Estimation of Hodoyoshi satellite AOS/LOS, and timing of data transmission (Continued)
 - All sensor are alloted time slot for data transmission to avoid crosstalk of radio waves (FATDMA: Fixed Access Time Division Multiple Access)
 - 10 sec mode: up to 5 sensors (12 sec time slot)
 - 1 sec mode: up to 50 sensors (1.2 sec time slot)
 - There are more than 50 sensors in the area where their transmitted radio waves can reach at the same time, additional satellites which has S&F capability are required. (Future plan)

Link Budget Analysis

- Value Unit Note Item • 1 W transmission Transmitted dBm 1W 30 power power is enough. **Received C/No** dB 43 -Communication Required C/No dB 36 distance: 1,000 Km Link margin dB 7 (600 km altitude, 30 deg elevation angle)
 - -Frequency: 400 MHz
 - -Gain for antennas (dipole):
 - -10 (ground) and 0 (satellite) dBi
- Reference: Store & Forward on-board satellite communication receiver for Hodoyoshi 3rd and 4th satellites, NE-G120004, April 23rd 2012, Next generation Space system Technology Research Association

Communication Link Analysis (1)

Pass	Day	AOS ¹ Time (UTCG)	LOS ² Time (UTCG)	Max Elevation (Deg)	Mean Range (km)	Duration (min:sec)
#1	1	8:28:40	8:32:16	52.5	930	3:36
#2	1	19:14:48	19:18:11	47.7	959	3:21
#3	2	8:41:12	8:43:37	36.6	1032	2:25
#4	2	19:26:27	19:30:27	69.8	908	4:00
#5	3	19:38:23	19:42:29	81.4	889	4:06
#6	4	19:50:34	19:54:17	55.7	914	3:43
#7	5	07:41:38	07:43:50	35.2	1046	2:12

Data transmission timing from a sensor located in Egypt to the Hodoyoshi satellite with 30-deg elevation constraint



Communication Link Analysis (2)

Satellite	Semi-major Axis (km)	Inclination (Deg)	Eccentricity
Hodoyoshi 3	7022 (644)	97.978	0.0035
Hodoyoshi 4	7014 (636)	97.980	0.0024

Timing of Hodoyoshi satellites flying over a sensor in Egypt



- Interval of data transmission:
 - Typical: 11 or 13 hours
 - Worst case: 24 hours

Communication Link Analysis (3)

	Ground Sensor in Egypt				$ \longrightarrow $		Ground St	tation in Jap	an
	Day	AOS Time (UTCG)	LOS Time (UTCG)	Duration (min:sec)	Latency (hour:min)	Day	AOS Time (UTCG)	LOS Time (UTCG)	Duration (min:sec)
#1	1	8:28:40	8:32:16	3:36	2:43	1	11:15:40	11:16:41	1:01
#2	1	19:14:48	19:18:11	3:21	5:16	2	00:33:46	00:37:26	3:40
#3	2	8:41:12	8:43:37	2:25	2:43	2	11:26:38	11:29:35.	2:58
#4	2	19:26:27	19:30:27	4:00	5:15	3	00:45:35	00:49:39	4:04
#5	3	19:38:23	19:42:29	4:06	5:15	4	00:57:33	01:01:41	4:07
#6	4	19:50:34	19:54:17	3:43	5:15	5	01:09:41	01:13:31	3:50
#7	5	07:41:38	07:43:50	2:12	4:18	5	12:01:59	12:06:05	4:06

Maximum data latency is less than <u>6 hours</u>.



Technical risks (1)

- Launch failure of Hodoyoshi 3rd and 4th
 - Hodoyoshi 2nd satellite can be used as back-up. (Launch date for Hodoyoshi 2nd is not fixed yet.)
 - In case of no available satellites, Limited observation activities will be performed with ground-base network
- Development failure: Space segment
 - Hodoyoshi 3rd and 4th satellites are currently in the phase of FM integration and testing without delay.
- Development failure: Ground segment
 - A prototype transmitter is being manufactured, and field test will be performed by the end of this year.

Technical risks (2)

- Development failure: Ground segment
 - Water resource observation sensor
 - Strict resource limitation
 - Power
 - Size
 - Data size
 - Cost
 - Maintenance/Calibration free
 - To minimize development risk, The most promising sensors based on priority, technology readiness level (TRL), cost, and availability will be adopted for this project.

Technical risks (3)

- One or more sensors not working properly or losing.
 - Replace only the sensor which is not working, not a whole system
 - Replaceable sensor design like PC accessories

Sustainability of the project

 International user community for water resource management will be organized.

–Egypt: MWRI/NWRC/NRI

–Japan: Japan Meteorological Agency–UNISEC Global



Organizations in Egypt (1)



عرتر اللومى ايحوث أعم

Government agency:

STREET, STREET



NARSS National Authority for Remote Sensing and Space Sciences

Organizations in Egypt (2)



Research center:



Prof. Mohamed Khalil Prof. Ayman Kassem Mr. Ahraf Nabil Rashwan

Dr. Abdelazim M. Negm



Future plan

- Development of S&F satellite constellation —Cubesats
 - Different algorithm to accommodate large number of sensors
 - -International cooperation
- Development of universal on-ground observation sensor
 - -Radiation monitoring, tracking of wild animals, tracing stolen objects, etc.

Schedule



Conclusion

- Our "Utilizing Nano Satellites for Water Monitoring for Nile River" is a very unique mission and it can positively impact global society, especially after the cubesat-size S&F satellite constellation is deployed.
- Since Hodoyoshi 3rd and 4th satellites are almost ready for launch, the technical feasibility is high and the technical risk involved is considered minimal.
- The future plan calls for the multinational collaboration. The harmonized effort by the international teams including Japan and Egypt is crucial to achieve the common and ambitious goals to contribute global society.

Thank you for your attention

This research is made possible by the grant from the Japan Society for the Promotion of Science (JSPS) through the "Funding Program for World-Leading Innovative R&D on Science and Technology (FIRST Program)," initiated by the Council for Science and Technology Policy (CSTP).