



Observation of Telomere Length Changes in the Deep Space Radiation Environment







NEED

Between 2000 and 2015:

• Life expectancy at birth increased 5.5 years.

• Healthy life expectancy increased 4.6 years

What are Telomeres?

Telomeres are located at the tips of chromosomes. In humans, the telomeric repeat sequence is 5' TTAGGG 3' and this important structure is responsible for protecting chromosomesduring DNA replication by providing additional length during the replication process. Telomerase is an enzyme formed by a ribonu cleic acid protein complex, present in germ cells, differentiated stem cells and fetal tissues. Telomeres are shortened with each replication and telomerase restores telomeres, maintaining DNA homeostasis (i.e., the stable and ideal condition of the organi sm) When telomeres are kept long, the chromosome is properly protected, ensuring correct replications, and reducing the effects of aging such as mutations and excessive apoptosis (cell death)





Telomere shortening related diseases

Table 1 is a list of some of the diseases related to telomere shortening.

Neurological diseases

Telomere shortening is related to the pathogenesis of neurological diseases and intellectual disabilities

- Autism Spectrum Disorders (ASD), Epilepsy and other intellectual disabilities: the telomeres, because of their short length causing the inability to protect the DNA well, lead to the mutation of the ACTL6B gene
- Schizophrenia: related to the gene SETD1 A, a lysine methyltransferase. Deletions (shortening of DNA and therefore loss of genetic material) of chromosome 22, also called 22g11.
- Alzheimer's disease (AD): due to the death of cells due to their accelerated aging, caused by very short telomeres.
- Parkinson's disease (PD).
- Depression.

 Cancer: cancer, also known as malignant neoplasm, is a disease caused by an uncontrolled multiplication of abnormal cells, which invade nearby tissues. DNA mutates due to telomere shortening, caused because of a deficit in telomerase production. A healthy cell would undergo the process of senescence (cellular aging) and lead to apoptosis (programmed cell death). Cancer cells evade this process by producing telomerase.

Genetic diseases

Related to DNA mutation, because the telomeres are

too short and do not produce enough telomerase to protect the genetic code.

- Dyskeratosis congenita: related to telomere shortening, which is unable to protect the DNA correctly, generating a mutation in the DKC1 gene.
- Aplastic anemia: the short telomere length in hematopoietic cells proceed to mutation.

Other diseases.

- Bone marrow failure: It is a hematopoietic dysfunction caused by defective and short telomere structure, causing DNA not to repair properly.
- Pulmonary fibrosis: This disease is generated due to poor telomerase segregation, leading to telomere shortening to avoid mutations. It has also been observed that cryogenic liver cirrhosis is related to the mutations accompanying this disease.



Existing work on telomere lengthening in space.

The NASA twins experiment was the first project to question the effects of zero gravity and space travel on cellular aging. The results showed that the twin who remained on Earth had no major changes in their telomeres. However, what was impressive was that the telomeres of the twin in space lengthened significantly. This experiment demonstrates that antigravity may work as a possible solution to the treatment of diseases caused by short telomeres.

Why is our experiment important?

Spirit of Inquiry aims to study telomere biology by observing their behavior in deep space. This basic understanding of the molecular mechanisms could lead to a better understanding of the general aging process in humans and the resulting diseases. For this mission we have decided to study the effects of high energy radiation on telomeres, as this environment is difficult to replicate on Earth for prolonged periods. To our knowledge, there is not enough scientific evidence and knowledge about telomeres being exposed to ioni zing radiation and its consequences in the long term. According to the National Cancer Institute, high energy radiation is extremely dangerous

since it has sufficient energy to eject electrons from atoms or molecules, especially nucleic acids. Ionizing rad iation produces chemical changes in cells and damages DNA which could eventually lead to cancer development. Studies have shown that exposure of telomeres to ionizing radiation increases telomerase activity (key to telomere lengthening so that DNA remains healthy), especially in cancer cells

Sustainable Development-goals



Health and well-being

Studying how telomeres behave in the deep space environment will improve our understanding of the mechanisms that affect aging and disease, which could impr ove quality of life and increase life expectancy.



Innovation

This mission will serve as a demonstrator of using proven technologies to quickly develop and run new experiments while accelerating transfer of existing technologies to emerging space organiza tions. This is important for increasing democratization of deep space.

Reduced inequalities

The knowledge gained can help develop affordable treatments to attenuate the effects of cancer, cellular aging, and prevention of neurodegenerative and autoimmune diseases, reducing inequalities derived from elderly care.



MISSION OBJECTIVES

GENERAL

Investigate the effects of deep space high energy radiation and microgravity on telomeres using yeast as a model organism.



SPECIFIC

Contribute to future human missions to Mars by deepe ning the understanding of the risks presented by cosmic rays and high energy solar radiation during deep space flight.

Contribute to regenerative medicine by observing telomere length changes in deep space to better understand the aging process, cell rejuvenation, and identify mechanisms of cancer growth.

OPERATIONS

Including orbital design experimental concept and setup

FIRST PHASE Pre laund

Pre launch operations

- The satellite is delivered to the launch provider 6 months in advance of the launch.
- The CubeSat is handed over to the integrator and the yeast remains dehydrated until launch.

THIRD PHASE Experiment operations

- Detumbling.
- Attitude correction and solar panel deployment.
- Rehydrated yeast

SECOND PHASE La

Launch and Coasting

- Launch as secondary payload.
- Coast for approximately 4 days after launch.

FOURTH PHASE

Mission disposal

• Remain in heliocentric orbit and transmit position.



ORBITAL DESIGN

Diagram of mission's orbit

- Heliocentric orbit.
- Upper stage disposal.

Maximum distance during mission from earth 0.3 AU and Sun 1.04 AU.

Perihelion 0.99 AU aphelion 1.05 AU.











The downlink bit rate ranges from 8 bit/s to 32 kbit/s. Uplink has two modes: 16 bit/s for emergency and 1000 bit/s for normal operations. The TT&C and communications sys tem is effective up to 0.4 AU. The maneuverable antenna is accurate to 1°.

KEY PERFORMACE PARAMETERS



The yeast will be supplied with nutrients and rehydrated after Spirit of Inquiry exits Earth's magnetosphere.



Each satellite will carry ten microfluidic cartridges. Each cartridge will contain eight unhealthy short telomere yeast samples, and eight health y normal telomere yeast samples.

Temperature within the research payload of 25C ± 1



Technologies that must be further developed:

Inserting fluorescent proteins (GFPs) next to telomere sequences:

The Technology Readiness Level of inserting fluorescent proteins (GFPs) next to telomere sequences is nine because it has been tested successfully, for example to study interactions of human telomeric DNA binding proteins with telomeric DNA

• Cold gas attitude thrusters:

The Technology Readiness Level of Cold gas attitude thrusters is nine because it has been tested in successful mission operations. The development of cold gas thrusters in CubeSats is not as advanced as in rockets, but there is resear ch in the development of 3D Printed Cold gas thrusters for the BioSentinel mission.





DESCRIPTION

OVERVIEW

0 6U

🗅 19kg

Solar powered

• Reaction wheels and thrusters

• 4U for experiment



EXPERIMENT EQUIPMENT

For the experiment, yeast will be used to study the behavior of DNA in deep space, specifically S. Castelli, because it is known for having regular telomere sequences (TCTGG-G(TG). This is crucial for simulating human telomeres. Inserting Green Fluorescent Proteins (GFPs) next to these sequences will allow the length of telomeres to be monitored. The proteins fluoresce when exposed to light of a specific wavelength, which is provide d by an LED. Photodiodes detect the fluorescent light and record the intensity, which corresponds to the length of the telomeres. These intensities will be compared to a control sample on Earth.

To reduce development time and risk, the yeast will be stor ed and grown with the system developed for NASA's Bio-Sentinel mission. Microfluidic cartridges will contain the yeast, with one cartridge having 16 wells. Eight wells will be filled with unhealthy yeast that have short telomeres, and the other eight are filled with healthy yeast with normal telomeres. The cartridges will allow for nutrient control and a stable growth environment. Heating elements will also be used for temperature stability. The yeast will be deprived of sugar and water until the spacecraft is released from the upper stage and stabilized. On earth, 10 microfluidic cartridges will be grown in the same system and monitored as the control. The onboard computer will capture the length measurements along with measurements of the radiation and send them back. On earth, 10 microfluidic cartridges will be grown in the same system, and monitored as the control 5.3

EQUIPMENT

- TT&C and communications system used by PROCYON.
- Two low gain X band patch antennas for uplink with 7.1 GHz of frequency
- Two low gain X band patch antennas for downlink with 8.4 GHz of frequency.
- One medium gain X band patch antenna, and one maneuverable high gain X band antenna.
- Bit rate varies from 8bps to 32kbps.
- High efficiency (30%) Gallium Nitride power amplifiers.

The power consumption of the system is 44 W. 60 min per pass to retrieve mission science, health and status, and performance data and to uplink commands.







- Emergency telemetry and high rate telemetry: the time delay for communications could be around 20 minutes.
- For ground operations, we will use the Ground Station for Deep Space Exploration and Telecommunication located in Saku City in Japan.





POSITION AND ATTITUDE CONTROL

- Star-tracker, IMU and sun sensors
- Reaction wheels for fine adjustments
- Cold gas thrusters for detumbling and desaturation
- 150g isobutane with Isp of 65 s

POWER GENERATION AND STORAGE:

• 73W generated from solar panels

• No day/ night equipment cycle

• Peak load 63 W

• Idle in experiment mode 21W

• 48.8 Wh of battery capacity



SPECIAL ASPECTS

| Estimated hardware costs | | | | | |
|---|--|--|--|--------------------|--|
| Research \$33,824 | Power: \$98,400 | Propulsion \$88,936 | Communications \$24,797 | Structural \$6,400 | |
| 1 S. Castelli yeast: \$490 1 Fluorescent Protein: \$294 1 Photodiode sensor:\$60 1 Temperature sensor:\$232 1 Radiation sensor:\$475 1 Pressure sensor:\$378 1 LEDs:\$185 | Solar Panels: \$9900 Batteries: \$28.15 | Reaction wheels: \$20,000 IMU: \$236 Cold Gas Thrusters \$ 20,000 Monopropellant thruster:\$13400 Star Tracker: \$32 000 Sun sensors:\$3300 | Low Gain Patch: \$8795 Antennas: \$1993 PSK modulation: \$4515 2 medium gain patch antenna:\$5394 X band patch antenna: \$4100 | 6U CubeSat | |



Estimated total costs

| Hardware Costs | Development costs | Launch costs | Operations costs | TOTAL |
|----------------|-------------------|--|------------------|--------------------------|
| \$252 357 | \$200,000 | It is approximately \$ 600 thousand dollars for 6 units, however CubeSats have not been sent com- mercially to deep space before. | \$40 000 | \$492,000 - \$1, 092,000 |

Project Implementation 2022 2024

| Design | 5 - 6 Months |
|-------------------------------------|--------------|
| Development (experiment & software) | 8 Months |
| Construction | 4 Months |
| Testing | 2 Months |
| Launch Vehicle Integration | 6 Months |
| Primary Mission | 8 months |
| | |



RISK MITIGATION

RISKS R&D fo Maximizing launch opportunities Rad Reducing likelihood of delays Reduce likelihood of component failure Increasing safety margins Unsuccessf

| or Telomere Observation Method | Medium | |
|--|---------------|--|
| Battery Degradation | Low | |
| diation Damage to instruments | Medium | |
| Yeast life support failure | Low | |
| Insufficient funding | High | |
| ful initial communication with satellite | Medium - High | |
| | | |





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